

## HANDLY & FREQUENTLY USED FORMULAS FOR THERMAL ENGINEERS

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### GEOMETRY & MATH | GEOMETRI & MATEMATIK

Cylindrical (Tube) Volume  $V = p / 4 \cdot d^2 \cdot L$  [m<sup>3</sup>]

Cylindrical (Tube) Surface  $A = p \cdot d \cdot L$  [m<sup>2</sup>]

Diameter  $d = \sqrt{4 \cdot A / p}$  [m]

Rectangular Triangle A = 90°: Geometrical Vector Sum

$$a^2 = b^2 + c^2 \quad \hat{U} \quad a = \sqrt{b^2 + c^2} \quad [\text{m}]$$

$$\cos B = c / a \quad ; \quad \sin B = b / a \quad ; \quad \tan B = b / c$$

## STRENGTH & STATICS | STYRKELÆRE & STATIK

Force  $\mathbf{F} = m \cdot a$  [N]

Stress  $\mathbf{s} = \mathbf{F} / \mathbf{A}$  [N/m<sup>2</sup>]

Stress  $\mathbf{s} = \mathbf{e} \cdot \mathbf{E}$  [N/m<sup>2</sup>] Hook's Law

Strain  $\mathbf{e} = \mathbf{DL} / \mathbf{L}$  [-]

A = Cross section area [m<sup>2</sup>]

E = Elasticity Modulus [N/m<sup>2</sup>]

m = Mass [kg]

a = Acceleration / Gravity Acceleration [m/s<sup>2</sup>]

$\Delta L$  = Deformation in Length [m] ; L = Length [m]

Bending Stress in beams  $\mathbf{s} = \mathbf{M}_T / \mathbf{W}$  [N/m<sup>2</sup>]

$M_T$  = Torque [Nm]

W = Section Modulus [m<sup>3</sup>] profile depending

**Simple Supported Beam – Uniform spread load**

Max. torque  $\mathbf{M}_{T, MAX} = \mathbf{P} \cdot \mathbf{L} / 8$  [N·m]

at middle of the beam

Max. Reflection  $\mathbf{U} = 5 \cdot \mathbf{M} \cdot \mathbf{L}^2 / (48 \cdot \mathbf{E} \cdot \mathbf{I})$  [m]

at middle

**Cantilever Beam – Uniform spread load**

Max. torque  $\mathbf{M}_{T, MAX} = \mathbf{P} \cdot \mathbf{L} / 2$  [N·m]

at the fixed support in the wall

Max. Reflection  $\mathbf{U} = \mathbf{M}_{T, MAX} \cdot \mathbf{L}^2 / (4 \cdot \mathbf{E} \cdot \mathbf{I})$  [m]

at free end of the beam

P = Total uniform load of beam [N]

I = Moment of Inertia [m<sup>4</sup>]

## HEAT & TEMPERATURES | VARME & TEMPERATUR

Absolute Temperature (Kelvin)

$$T = t + 273,15 \text{ [K]}$$

t = Temperature [°C]

Heat / Heat Content  $Q = m \cdot C_p \cdot (t_2 - t_1)$  [W] | [J]

m = Mass Flow [kg/s] / Mass [kg]

$C_p$  = Specific Heat [J/(kgK)]

$t_1$  and  $t_2$  = Temperatures Inlet and Outlet [K] | [°C]

Linearly Heat Expansion of Materials

$$DL = L \cdot \alpha_L \cdot Dt \text{ [m]}$$

Volumetric Heat Expansion of Materials

$$DV = V \cdot \beta_V \cdot Dt \text{ [m}^3\text{]}$$

L = Length [m] ; V = Volume [m<sup>3</sup>]

$\alpha_L$  = Length Expansion Coefficient [1/°C]

$\beta_V$  = Volume Expansion Coefficient [1/°C]

$\Delta t$  = Temperature Change [°C]

For Ideal Gasses :

$$p \cdot v = R \cdot T = p_0 \cdot v_0 \cdot (1 + t / 273,15)$$

Specific Volume  $v = 1 / \rho$  [m<sup>3</sup>/kg]

p = Pressure (bar abs.) ;  $\rho$  = Density [kg/ m<sup>3</sup>]

T = Absolute Temperature [K]

$p_0 \cdot v_0$  : Pressure and Specific volume at 0°C

R = Gas Coefficient [J/(kg·K)] :

Air = 287,1 J/(kg·K) Steam = 461,5 J/(kg·K)

1 kmol equals a volume of 22,4138 m<sup>3</sup>

$$m = n \cdot M \text{ [kg]}$$

$V_n = n \cdot V_{mol}$  [m<sup>3</sup>] at 0°C and 1,01325 bar

$$\rho = m / V_n \text{ [kg/m}^3\text{]}$$

M = Mol mass [kg/mol] ;  $\rho$  = Density [kg/m<sup>3</sup>]

$V_n$  = Normal Volume [m<sup>3</sup>] ; n = Number of mol

$V_{mol}$  = Molar Volume [m<sup>3</sup>/mol] ; m = mass [kg]

HEAT TRANSFER | VARMEOVERFØRING

BY CONVECTION | VED KONVEKTION

Heat Transfer by Convection  $Q = k \cdot F \cdot DJ$  [W]  
F = Heat Surface – Total wall area [m<sup>2</sup>]

Heat Transmission Coefficient

$$k = 1 / (1/a_1 + 1/a_2 + e/l + f_1 + f_2) \text{ [W/(m}^2\cdot\text{K)]}$$

$\alpha_1$  = Heat Transfer Coefficient – Fluid 1 [W/(m<sup>2</sup>·K)]

$\alpha_2$  = Heat Transfer Coefficient – Fluid 2 [W/(m<sup>2</sup>·K)]

$\lambda$  = Heat Conductivity Wall Material [W/(m·K)]

e = Wall Thickness [m]

$f_1$  = Fouling Coefficient – for the wall of fluid 1 [m<sup>2</sup>·K/W]

$f_2$  = Fouling Coefficient – for the wall of fluid 2 [m<sup>2</sup>·K/W]

BY RADIATION | VED STRÅLING

Radiation Heat between two surfaces 1 and 2

$$F = C_{12} \cdot F_1 \cdot ((T_1/100)^4 - (T_2/100)^4) \text{ [W]}$$

Radiation Coefficient

$$C_{12} = 1 / (1/C_1 + 1/C_2 - 1/C_s) \text{ [W/(m}^2\cdot\text{K)]}$$

$$C = e \cdot C_s \text{ [W/(m}^2\cdot\text{K)]}$$

$\varepsilon$  = Emission ratio [-]

$C_s$  = Radiation Coefficient for the absolute black surface [-]

T = Absolute temperature [K]

Logarithmic Middle Temperature Difference

$$DJ = (Dt_1 - Dt_2) / \ln (Dt_1/Dt_2) ; \text{ all values in [K] | [}^\circ\text{C]}$$

$\Delta t_1$  = Difference in Temperatures of Fluid1 and Fluid 2 at “1”

$\Delta t_2$  = Difference in Temperatures of Fluid1 and Fluid 2 at “2”  
“1” and “2” being the physical positions of the inlets and outlets of heat exchanger in current or counter flow types

Nusselt's Number

$$Nu = a \cdot L_F / \lambda \quad [-] \quad \hat{U} = a \cdot L_F / L_F$$

$\alpha$  = Heat Transfer Coefficient [W/(m<sup>2</sup>·K)]

$L_F$  = Flow Length [m] e.g. diameter or plate length

$\lambda$  = Heat Conductivity Fluid [W/(m·K)]

General expression for forced circulation

$$Nu = K_1 \cdot Re^{K_2} \cdot Pr^{K_3}$$

General expression for natural circulation

$$Nu = K_5 \cdot Gr^{K_4} \cdot Pr^{K_3}$$

Prandtl's Number  $Pr = r \cdot C_p \cdot n / \lambda \quad [-]$

Grashoff's Number  $Gr = g \cdot r \cdot DV \cdot Dt \cdot L_F^3 / n \quad [-]$

g = Gravity acceleration [m/s<sup>2</sup>]

$K_1, K_2, K_3, K_4$  and  $K_5$  are different constants and equations based on tests and depending on the type of heat transfer.

## MECHANICS OF FLUIDS | STRØMNING & VÆSKEFYSIK

Total pressure  $p_T = p_s + p_D$  [N/m<sup>2</sup>]

Dynamic pressure  $p_D = \frac{1}{2} \cdot c^2 \cdot r$  [N/m<sup>2</sup>]

Pressure Height  $p_H = g \cdot r \cdot H$  [N/m<sup>2</sup>]

$p_s$  = Static pressure [N/m<sup>2</sup>]

$g$  = Gravity acceleration [m/s<sup>2</sup>]

$H$  = Height / Altitude [m]

Bernoulli's Law about constancy in pressure. All in [N/m<sup>2</sup>]

$$p_{s,1} + p_{D,1} + p_{H,1} = p_{s,2} + p_{D,2} + p_{H,2} \quad \hat{U}$$
$$p_{s,1} + \frac{1}{2} \cdot c_1^2 \cdot r + g \cdot r \cdot H_1 = p_{s,2} + \frac{1}{2} \cdot c_2^2 \cdot r + g \cdot r \cdot H_2$$

For Ideal Gasses:

Dynamic Viscosity  $h @ h_0 \cdot (273 + t) / 273$  [N·s/m<sup>2</sup>]

$t$  = Temperature [°C]

Dynamic Viscosity  $h = n \cdot r$  [Pa·s] | [kg/(m·s)]

Reynold's Number  $Re = c \cdot L_F / \nu$  [-]

$\nu$  = Kinematic Viscosity [m<sup>2</sup>/s];  $\rho$  = Density [kg/m<sup>3</sup>]

$c$  = Velocity [m/s];  $L_F$  = Flow Length [m]

Pressure Drop in tube  $\Delta p_{TB} = \lambda \cdot p_D \cdot L_T / d$   
 $= \lambda \cdot \frac{1}{2} \cdot r \cdot c^2 \cdot L_T / d$  [N/m<sup>2</sup>]

$\lambda$  = Friction Coefficient [-];  $L_T$  = Tube Length [m]

$d$  = Internal Tube Diameter [m];  $\rho$  = Density [kg/m<sup>3</sup>]

$c$  = Velocity [m/s]

Pump Capacity  $P = h_T \cdot q_V \cdot \Sigma \Delta p$  [W]

Total Efficiency  $h_T = (h_{PUMP} \cdot h_{MOTOR})$

Efficiency  $h = P_{PERFORMED} / P_{ABSORB}$

$q_V$  = Volume flow [m<sup>3</sup>/s]

$\Sigma \Delta p$  = Sum of all pressure drops in the circuit [Pa]

## ELECTRICITY | ELECTRICITET

Power / Capacity of a 1-Phase System :

$$P = U_{PH} \cdot I_{PH} \text{ [W]}$$

Power / Capacity of a 3-Phase System :

$$P = \sqrt{3} \cdot U_N \cdot I_N \cdot \cos \varphi \text{ [W]}$$

$U_N$  = Net Voltage [V] ;  $I_N$  = Net Current [A]

$U_{PH}$  = Phase Voltage [V] ;  $I_{PH}$  = Phase Current [A]

$\cos \varphi$  = Phase Angel [-]

$\cos \varphi = 1$  for Heating elements and other simple resist.

$\cos \varphi < 1$  for Electrical Motors (inductive resistance).

Power, Voltage and Current in Conventional Resistances

Ohm's Law

$$U = R \cdot I \text{ [V]}$$

Power expressed by the resistance

$$P = U \cdot I = U^2 / R = I^2 \cdot R \text{ [W]}$$

$U$  = Voltage [V] ;  $I$  = Current [A]

$R$  = Resistance [ $\Omega$ ] | [Ohm]